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TECHNICAL REPORT

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TERMINAL BALLISTIC STUDY OF FLECHETTES (U)

BY

C. A. RIDDLE

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REPORT NO.: WAL TR 768.1/1(c)
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DECEMBER 1958

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CANISTER AMMUNITION,
FLECHETTE

TERMINAL BALLISTIC STUDY OF FLECHETTES (U)

TECHNICAL REPORT

By

C. A. Riddle

O.O. Project: TV-121, Long Range Development of
Mass Counter-Assault Ammunition
D/A Project: SA74-01-002
Report No.: WAL TR 768.1/1(c)
Filing Subject: Canister Ammunition, Flechette

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TITLE

TERMINAL BALLISTIC STUDY OF FLECHETTES (U)

ABSTRACT

✓ Army Vgo ballistic limits were determined with the FL-17, FL-17C-4, 10, 12, 14, and 16 flechettes against aluminum targets at 0° obliquity; and with the FL-17, FL-17D-6, and FL-17C-12 and 16 flechettes against aluminum targets at 45° obliquity.

Employing the ballistic limit data, deMorrow-type equations were derived for the flechette designs and obliquities studied. A satisfactory reproduction of the ballistic data within a maximum error of 5% was obtained with the equations derived. The order of superiority of flechettes against the aluminum targets evaluated was ~~stabilized~~ *stabilized* K

0° Obliquity

1. FL-17
2. FL-17C-10, 12, 14, and 16 (all equal)
3. FL-17C-4

45° Obliquity

1. FL-17
2. FL-17D-6, FL-17C-12 and 16 (all equal)

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INTRODUCTION

As part of a continuing effort to develop improved caliber ammunition for employment against massed infantry assaults, such as encountered in Korea, the Ordnance Corps has been investigating flechette-type missiles of various designs. Numerous designs have been evaluated for flight stability, range-velocity characteristics, quantity and ease of loading, economy of fabrication, effect of manufacturing deviations on flight performance, etc., by Ordnance Corps contractors under Contracts DA-33-008-ORD-160, DA-33-008-ORD-1257, and DA-33-008-ORD-1562. As part of this program, Watertown Arsenal had been requested to conduct terminal ballistic studies with various flechette designs against lightweight armor materials, such as employed in armored vests, helmets, and personnel carriers. Materials chosen for this study were unbonded nylon, aluminum, and Hadfield-manganese steel. These materials were evaluated with the FL-17 (8-grain) flechette at 0° and 45° obliquity and the resulting data published. It was concluded that unbonded nylon, and probably all fabric materials that fail to deform the missile, offers very little ballistic resistance to flechette attack, and that Hadfield-manganese steel, because of a regrettable variation in ballistic performance of similar plate, is not suitable for distinguishing variations in ballistic performance of individual flechette designs. Accordingly, it was recommended that future terminal ballistic flechette studies be conducted with aluminum as the target material. This report covers these tests and also incorporates the data previously gathered with flechettes against aluminum armor. Of the seven flechette designs studied it is possible to differentiate between penetration performance against the aluminum armor material and to select the missile that exhibits optimum performance.

It was deemed desirable to extend the ballistic limit data to higher velocity levels, however, limitations imposed by structural failures of the various flechette designs during set-back in launching from the gun, restricted the upper velocity limits that could satisfactorily be employed with each design.

MATERIALS

Flechettes

The following seven flechette designs were employed to determine ballistic limit data for comparisons of penetration performance:

Type	Weight
1. FL-17C-4	4 Grains
2. FL-17D-6	6 Grains
3. FL-17	8 Grains
4. FL-17C-10	10 Grains
5. FL-17C-12	12 Grains
6. FL-17C-14	14 Grains
7. FL-17C-16	16 Grains

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These flechettes are illustrated in Figure 1 together with a tabulation of mean dimensions. In order to utilize the previous data, accumulated with the FL-17 flechette, firing was continued with this design in preference to the later FL-17D-8. It has since been determined that the newer FL-17D-8 design, which differs from the FL-17 in nose contour and a .605" larger diameter across the fins, exhibits superior velocity-dummy performance and greater depth of penetration into colorox targets, particularly in the super-sonic velocity regions.

Targets

Because of the previously noted limitations imposed by unhardened nylon and Hadfield-manganese steel as a means of differentiating between ballistic performance of the various flechettes, 2024-T4 aluminum alloy was employed as the target material. Army V50 ballistic limits were determined at 0° and 45° obliquity against the target thicknesses tabulated in Table I.

PROCEDURE

Method of Measurement

In order to evaluate the arm penetration capabilities of the various flechette designs a number of rounds of each were fired at the aluminum targets and Army V50 ballistic limits determined. A sufficient number of rounds were fired so that projectile defeat and penetration of the target were achieved over the velocity range traversed. According to the Army criteria, a complete penetration of the armor occurs when a hole or crack on the reverse side of the plate, caused by the missile impacting the target, is sufficient to permit the passage of light. The V50 Army limit is that velocity at which a 50% probability exists that the missile will achieve a complete penetration of the armor. The results from the ballistic test plate firings are incorporated in the following formula for the calculation of the V50 ballistic limit:

$$V_{50} = \frac{\Sigma V + K(NP - NC)}{NP + NC}$$

and

$$K = \frac{V_{mp} - V_{LC}}{2}$$

and if

$$V_{LC} > V_{mp}$$

then

$$K = 0$$

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where

V_{MP} = Highest velocity resulting in a partial penetration

V_{LC} = Lowest velocity resulting in a complete penetration

$V_{MP}-V_{LC}$ = Zone of mixed results

ΣV = Sum of velocities within the zone of mixed results including V_{MP} and V_{LC}

NP = Number of partial penetrations within the zone of mixed results including V_{MP}

NC = Number of complete penetrations within the zone of mixed results including V_{LC}

Flechette velocities were determined by their passage through a pair of lumline screens spaced 10 feet apart and connected to a 400 Kc counter-chronograph. The first screen activated and the second screen stopped the chronograph, thus permitting the determination of the missile velocity at a point midway between the screens located at a distance of 7.5 feet from the target. Since comparisons of individual flechette ballistic limits are made relative to this velocity, corrections were not applied for velocity loss from the midpoint of the lumline screens to the target.

Penetration Equations

A plot on log-log paper of Army V_{50} ballistic limit versus the caliber thickness ratio yields the curves shown in Figures 2 and 3. The slope of the straight line that most closely fits the data then corresponds to the exponential function $n/2$, and the intercept at $s/d = 1$ yields the constant term K in the following equation:

$$V_L = K \left(\frac{s}{d} \right)^{\frac{n}{2}},$$

which is derived from a deMarre penetration equation that has previously been described in detail.^{1,3,4,5}

The exponent $n/2$ in the foregoing equation very often permits an interpretation of the mechanism of armor penetration dependent upon projectile nose shape and armor hardness. In the case of flechette penetration, however, the exponents are quite different from those normally encountered in deMarre equations and indicate mechanisms of penetration of a different type than normally observed for kinetic energy armor piercing projectiles. Thus, the equations developed for the different flechettes are presented only as a convenient method of representing the ballistic test results and apply only to aluminum armor material with the flechettes studied.

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RESULTS AND DISCUSSION

A summary of the ballistic test data is presented in Table I, and the penetration equations derived from these data are contained in Table II. Presented in Table III are the actual ballistic limits obtained (V_{50}) and a comparison with those computed from the penetration equations (V_p). Close agreement is demonstrated, the average deviation for 50 ballistic limits being 2.5% between the actual and derived limits. Examination of Figures 2 and 3 demonstrates that the FL-17 flechette defeats the aluminum targets at lower velocities than the remaining flechettes. The light, FL-17C-4 flechette is inferior to the FL-17 at 0° obliquity by approximately 500 ft/sec over the entire range of target thicknesses tested. The heavier missiles, FL-17C-12, 12, 14, and 16 all demonstrate similar performance and are considerably inferior to the FL-17 against the thinner targets, while approaching the FL-17 in performance against the heavier targets at an s/d ratio of 3.5. Above this point inferior performance is again manifested for the FL-17C-12 missile in the only two cases tested with heavier flechettes. At 45° obliquity superior performance over the entire range of targets evaluated is demonstrated for the FL-17 design over the FL-17D-6, FL-17C-12, and 16, the superiority increasing slightly as the armor thickness increases.

RECOMMENDATIONS

Because of the superior performance of the FL-17 flechette it should be selected in preference to the other designs tested for canister filler. However, before establishing the FL-17 as standard filler, similar terminal ballistic studies should be conducted with the FL-17D-6 flechette to determine which of these two designs exhibit optimum penetration performance against aluminum targets.

Use of these flechette designs either lighter or heavier in weight than the FL-17 series is not recommended since no advantage of terminal performance is gained with these missiles, and only in the lighter designs is there any advantage gained in density of loading.

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TABLE I
SUMMARY OF PENETRATION DATA
Flechette vs 2024-T4 Aluminum

0° Obliquity								
Flechette	•	BL	HP	IC	20P	HP	NC	Rounds Fired
FL-17C-4	.007	1428	1355	1210	345	8	7	31
	.136	2048	2300	1890	410	8	9	30
	.193	2423	2540	2220	320	10	10	34
	.250	2771	2800	2570	290	7	10	34
FL-17	.100	973	1050	905	755	7	18	44
	.120	1094	1075	1005	70	4	5	50
	.150	1412	1470	1355	65	7	1	63
	.172	1517	1565	1490	75	5	0	45
	.226	1864	2000	1460	800	21	13	80
	.312	2400	2460	2140	350	21	21	50
	.350	2517	2540	2420	180	4	5	15
	.400	2982	3030	2900	70	6	3	19
	.460	3261	3310	3220	80	5	1	20
FL-17C-10	.001	1502	1615	1500	200	12	10	35
	.100	1753	1775	1650	115	12	5	21
	.185	2023	2280	1900	260	12	20	36
	.252	2141	2300	2030	270	11	14	23
FL-17C-12	.001	1463	1500	1450	70	2	7	20
	.103	1821	2080	1820	360	10	11	21
	.169	1830	1840	1740	290	11	9	26
	.263	2100	2100	2010	170	11	0	32
	.312	2100	2250	2040	210	10	6	30
	.330	2367	2400	2250	25	3	4	13
	.400	2600	2725	2560	10	5	2	33
	.460		3445					
FL-17C-14	.001	1290	1305	1255	50	0	4	37
	.100	1830	1860	1865	85	14	4	39
	.191	1780	1850	1800	150	14	14	35
	.240	1945	2050	1930	130	10	13	31
FL-17C-18	.000	1275	1415	1270	145	14	7	30
	.101	1635	1745	1610	125	10	15	36
	.180	1800	1750	1650	80	10	11	30
	.240	1900	2010	1730	230	17	14	40
	.312	2182	2300	2135	85	3	4	18
	.35	2330	2330	2170	0	1	1	18

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TABLE I (Continued)

45° Obliquity								
Flechet to	o	BL	HP	LC	MC	NP	NC	Sample Fired
FL-17D-8	.180	2040	2040	2425	180	7	8	20
	.185	2055	2055	2435	200	13	10	21
	.190	2070	2070	2445	220	11	11	20
	.317	2085	2085	2455	240	10	4	15
FL-17	.180	1920	2010	2445	240	7	7	20
	.185	1945	2030	2455	260	16	8	20
	.190	1970	2045	2465	280	6	6	25
	.195	2000	2070	2480	300	10	8	20
	.202	2030	2100	2510	320	17	7	25
	.310	2045	2120	2520	340	6	5	25
FL-17C-14	.180	2340	2300	2250	170	8	8	24
	.185	2350	2315	2260	190	10	10	21
	.235	2400	2355	2300	210	8	7	23
	.314		2345					
FL-17C-18	.180	2160	2075	2090	185	8	10	20
	.185	2180	2100	2115	205	17	11	20
	.260	2215	2125	2135	225	16	10	20
	.317	2240	2150	2160	245	4	5	15

NOTE:

- * Target Distance (feet)
- * Angle of Obliquity (deg)
- * Angle of Elevation (deg)
- * Angle of Azimuth (deg)
- * Angle of Tilt (deg)
- * Angle of Roll (deg)
- * Angle of Yaw (deg)
- * Angle of Pitch (deg)
- * Angle of Heave (deg)
- * Angle of Sway (deg)
- * Angle of Twist (deg)
- * Angle of Bend (deg)
- * Angle of Curve (deg)
- * Angle of Turn (deg)
- * Angle of Swing (deg)
- * Angle of Rock (deg)
- * Angle of Lean (deg)
- * Angle of Tilt (deg)
- * Angle of Roll (deg)
- * Angle of Yaw (deg)
- * Angle of Pitch (deg)
- * Angle of Heave (deg)
- * Angle of Sway (deg)
- * Angle of Twist (deg)
- * Angle of Bend (deg)
- * Angle of Curve (deg)
- * Angle of Turn (deg)
- * Angle of Swing (deg)
- * Angle of Rock (deg)
- * Angle of Lean (deg)

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TABLE II
DERIVED PENETRATION FORMULAS
Fiechettes vs Aluminum

Fiechette Design	Chillquity (Degrees)	Range of Target Thickness (Lachme)	No. of Targets Tested	Equation	Maximum Deviation (%)
FL-17C-4	0	.007-.230	4	$V_L = 1100(\frac{V}{V_L})^{.731}$	0.0
FL-17	0	.100-.400	0	$V_L = 720(\frac{V}{V_L})^{.500}$	0.1
FL-17C-10	0	.001-.252	4	$V_L = 1340(\frac{V}{V_L})^{.307}$	0.1
FL-17C-12	0	.001-.450	0		
FL-17C-14	0	.001-.240	4		
FL-17C-10	0	.000-.310	0		
FL-17D-2	45	.150-.312	4		
FL-17C-12	45	.154-.316	4	$V_L = 1570(\frac{V}{V_L})^{.044}$	5.5
FL-17C-10	45	.150-.2125	4		
FL-17	45	.100-.310	0	$V_L = 1470(\frac{V}{V_L})^{.572}$	5.9

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TABLE III
COMPARISON OF MEASURED AND CALCULATED BALLISTIC LIMITS

0° Obliquity						
Flarehole	"	a/d	V_{50}	V_L	Δv	% Deviation $\left(\frac{\Delta v}{V_L}\right) \times 100$
FL-17C-4	.097	1.40	1432	1432	- 0	-0.0
	.149	2.25	2048	2048	- 17	-0.8
	.785	9.18	2425	2428	- 3	-0.1
	.988	4.69	2771	2755	+ 16	+0.6
FL-17	.100	1.289	972	969	+ 33	+3.4
	.120	1.87	1084	1088	- 24	-3.2
	.156	2.17	1418	1340	+ 72	+5.1
	.182	2.57	1517	1329	- 8	-0.5
	.250	3.87	1984	1887	- 91	-4.5
	.312	5.33	2400	2250	+ 150	+6.1
	.355	6.98	2517	2379	- 18	-2.3
	.400	8.38	2702	2580	-112	-3.7
FL-17C-10	.091	1.14	1509	1410	+ 92	+6.1
	.136	1.99	1753	1739	- 2	-0.1
	.195	2.64	2092	1949	+123	+6.0
	.282	3.83	2141	2119	+ 21	+1.4
FL-17C-12	.091	1.09	1483	1499	- 13	-3.0
	.163	1.94	1821	1745	+ 76	+4.2
	.189	2.25	1898	1850	- 20	-1.1
	.252	2.96	2108	2009	- 46	-2.3
	.319	3.73	2183	2250	- 67	-3.0
	.350	4.17	2387	2369	+ 7	+0.3
FL-17C-14	.091	.98	1290	1291	- 36	-2.8
	.189	1.79	1830	1848	- 30	-1.6
	.191	2.56	1740	1795	- 35	-1.4
	.246	3.43	1944	1979	- 38	-1.4
FL-17C-16	.099	.98	1275	1289	- 12	-1.0
	.161	1.77	1858	1887	- 35	-1.9
	.199	2.94	1860	1778	- 96	-5.0
	.248	2.87	1900	1880	- 20	-1.1
	.318	3.38	2161	2189	- 6	-0.3
	.35	3.78	2339	2287	+ 50	+2.2

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TABLE III (Continued)

45° Obliquity						
Flechette	ϕ	a/d	V_{50}	V_L	ΔV	% Deviation $\left(\frac{V_{50}}{V_L} - 1\right) \times 100$
FL-17D-S	.150	3.19	3695	2009	- 90	-3.0
	.180	3.04	3625	2835	-110	-3.0
	.220	3.47	3575	3482	- 83	+2.3
	.312	4.30	3655	4630	- 75	-1.0
FL-17	.100	1.20	1822	1778	+ 49	+3.7
	.130	1.67	1913	1870	+ 37	+2.0
	.160	2.17	2006	2280	+ 18	+0.7
	.180	2.61	2199	2546	- 45	-3.0
FL-17C-13	.202	3.08	2954	3620	- 70	-2.0
	.316	4.30	3448	3585	+ 63	+1.0
FL-17C-13	.150	1.08	2240	2380	- 90	-0.9
	.180	1.25	2783	2845	-140	+5.3
	.250	1.98	3099	3170	- 71	-3.3
	.314	3.74	3346MP			
FL-17C-16	.150	1.71	2108	2216	- 88	-9.5
	.187	1.81	2395	1480	- 95	-3.7
	.250	2.60	2815	2870	-155	-5.5
	.312	3.28	3448	3435	+ 15	+0.4

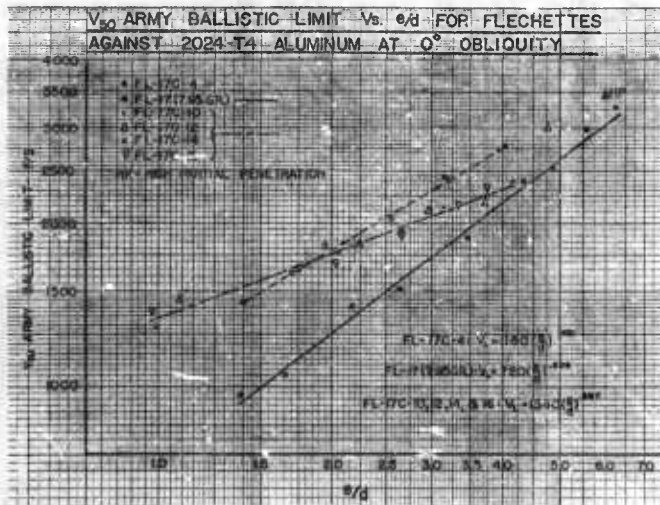
SYMBOLS:

- ϕ = Target thickness (inches)
- a/d = Ratio of target thickness to flechette diameter
- V_{50} = Measured array ballistic limit (ft/sec)
- V_L = Calculated ballistic limit (ft/sec)
- ΔV = ($V_{50} - V_L$)
- MP = Highest partial penetration

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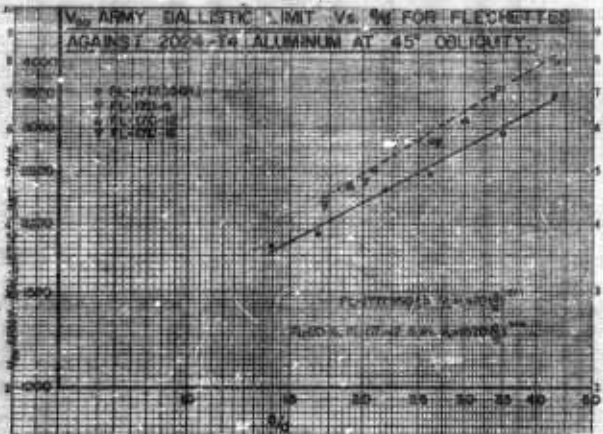
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FIGURE 2



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FIGURE 3

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